

MULTISCALE ANALYSIS OF ANISOTROPIC MATERIALS WITH HEXAGONAL MICROSTRUCTURE AS MICRO-POLAR CONTINUA

N. Fantuzzi,^{1,*} P. Trovalusci,² & R. Luciano³

¹DICAM Department, University of Bologna, Viale del Risorgimento 2, 40136, Bologna, Italy.

²DISG Department, Sapienza University of Rome, Via A. Gramsci, 53, 00197, Roma, Italy.

³Engineering Department, Parthenope University, Centro Direzionale (Isola C4), 80133, Napoli, Italy.

*Address all correspondence to: N. Fantuzzi, DICAM Department, University of Bologna, Viale del Risorgimento 2, 40136, Bologna, Italy., E-mail: nicholas.fantuzzi@unibo.it

This work discusses the advantages of micropolar theory in modeling anisotropic composite materials with microstructure. A homogenized constitutive model starting from a representative volume element is proposed in order to find an equivalent continuum. Classical (e.g. Cauchy of Grade 1) continua are not always suitable to accurately approximate the behavior of such composites because no size effects, nor lack of symmetries in strain and stress can be taken into account. This study focuses on composites made of hexagonal rigid particles which interact among themselves through elastic interfaces, so that the deformation energy of the material is concentrated only at the interfaces. Three particle geometries are investigated such as orthotetragonal, auxetic and chiral. Novel results have been achieved by presenting the behavior of panels with various material symmetries and subjected to concentrated loads.

KEY WORDS: multiscale, anisotropic materials, finite element method, Cosserat (micropolar), homogenization, RVE

1. INTRODUCTION

Composite materials can be studied by modeling interactions among their constituents or by homogenizing an equivalent continuum. The former approach generally requires higher computational cost because of the detailed modeling of particle/matrix interactions such as discrete element modeling (Baraldi et al. (2018); Reccia et al. (2018); Yang

1 et al. (2010)). The latter, as any field theory, is more efficient but its effectiveness is strongly related to the continuum
2 theory used and the homogenization method adopted to convert the physical particle/matrix system into an equivalent
3 continuum (e.g. Budiansky (1965); Ehlers (2011); Nemat-Nasser et al. (2013)).

4 In order to model complex spatial interaction effects or describe materials in which internal length scales are
5 not negligible when compared to structural length scales, homogenization techniques of different kind, have also
6 been extended to non-classical continua (see the review in Trovalusci (2014)). This latter circumstance becomes
7 significant when dealing with complex constitutive behaviors dependent on the microstructure size, such as strain
8 localization, and the field equations of the simple (Grade 1) classical continuum become ill-posed. To this regard,
9 some models include extra parameters, such as internal length, aiming to take into account the material nonlocality at
10 the constitutive level only without modifying the classical kinematics have been proposed by Alibert and Della Corte
11 (2015); Civalek et al. (2010); Demir and Civalek (2013); Larsson and Zhang (2007). Non-local or higher order
12 deformation gradient descriptions, specifically addressed to multiscale computational homogenization, have been also
13 proposed by Bacigalupo and Gambarotta (2010); Kouznetsova et al. (2004, 2002); Leismann and Mahnken (2015);
14 Massart et al. (2007); Peerlings and Fleck (2004); Sluys et al. (1993), as well as non-local explicit solutions obtained
15 for specific cases of elastic composites Bacca et al. (2013); Drugan and Willis (1996); Luciano and Willis (2000);
16 Smyshlyaev and Cherednichenko (2000). On the other hand, within the framework of 'implicit' non-local theories
17 concerning models with additional degrees of freedom Trovalusci (2014); Tuna et al. (2019); Tuna and Trovalusci
18 (2019), micromorphic continua, in particular continua with rigid local structure (micropolar), have been satisfactorily
19 applied to various composites Addressi and Sacco (2012); Bouyge et al. (2001); Forest et al. (1999, 2001); Forest and
20 Sab (1998); Masiani and Trovalusci (1996); Onck (2002); Ostoja-Starzewski et al. (1999); Tekolu and Onck (2008);
21 Trovalusci and Masiani (2003, 2005); Trovalusci and Sansalone (2007).

22 All the above mentioned models can be defined as non-local as the field equations contain internal length pa-
23 rameters revealing the presence of a hidden micro-structure which can affect the macroscopic behavior and there are
24 dispersion properties in wave propagation Kunin (1968). Size effect on such non-local materials can be investigated
25 as parametric simulations in order to better fit experimental evidences.

26 Homogenization based techniques have been widely exploited to study material failure behavior. For instance
27 Jain and Ghosh (2009) presented the damage evolution of composite materials via a Continuum Damage Mechanics
28 Model, Altenbach and Sadowski (2014) worked on failure and damage analysis of advanced materials. Implemen-
29 tation and computational aspects of multiscaled cracked problems have been proposed by Nguyen et al. (2012).
30 Collapse of three-dimensional systems made of blocks using a nonlinear implementation has been presented by Yang

1 et al. (2000). Analogously, Greco et al. (2016, 2017, 2018) studied the effects of micro-fractures and contact simula-
2 tions on the macroscopic response of elastic bodies in finite deformations and fiber reinforced materials.

3 When classical kinematics is enriched with extra degrees of freedom homogenization procedures have been
4 shown to provide more reliable models than in the case of classical local continua. As for instance in the study of
5 wave propagation within the framework of generalized continuum formulations for composite microcracked bars as
6 reported by Trovalusci et al. (2010) or also in the study of multiphysics problems such as thermo-elastic multifield
7 materials in Favata et al. (2016). In particular, micropolar theory introduces as degree of freedom the microrotation,
8 that is an additional kinematic feature of the material point, different from the local rigid rotation (the classical macro-
9 rotation), the rotational feature of the infinitesimal neighborhood, and their effects have been widely investigated by
10 Fantuzzi et al. (2019); Pau and Trovalusci (2012); Trovalusci and Masiani (2003, 2005); Trovalusci and Pau (2014)
11 for masonry-like materials. Note that these rotations coincide in the couple-stress theory, as well as in the classical
12 theory (see Masiani and Trovalusci (1996), Appendix). As it has been also recently analyzed by Fantuzzi et al. (2018),
13 micro-polar effects become prominent when geometrical or load singularities are present in the reference problem,
14 such as concentrated loads, voids or material inclusions and these effects have been also compared to those of other
15 kind of non-local continuum descriptions Tuna et al. (2019); Tuna and Trovalusci (2019).

16 The present work aims at presenting the mechanical micropolar behavior of hexagonal lattices (Rizzi et al. (2019);
17 Trovalusci et al. (2017)) with elastic interfaces. Different selection of hexagonal geometries and interface orientations
18 lead to a distinct material symmetry (Eremeyev and Pietraszkiewicz (2016)). In the present work three assemblies are
19 considered such as regular, hourglass and asymmetric placement of the hexagonal particles which are all derived from
20 a hexagonal pattern with sides of all the same lengths (equilateral). The first pattern regards regular hexagonal shapes
21 having the orthotetragonal symmetry The second is obtained from regular hexagons but with re-entrant corners, this
22 give an hourglass shape and auxetic properties to the material due to an equivalent negative Poisson ratio. Finally, the
23 third, is an asymmetric pattern given by one re-entrant corner and one not re-entrant starting from the regular shape.
24 The equivalent continuum in this case results to be chiral, by coupling the classical part of stress/strain with micro-
25 polar couple stresses/curvatures. Novel results have been achieved by presenting the behavior of the aforementioned
26 material configurations under concentrated loads.

27 This paper is structured as follows. First, Cosserat continuum model is briefly presented in order to introduce cur-
28 rent quantities and symbols. Second, a parametric hexagonal geometry based on four parameters is presented and the
29 investigated patterns are shown and defined. Third, the present in-house finite element implementation is presented
30 using linear finite elements with reduced integration. Finally, numerical applications are discussed by comparison

1 among the three considered geometries and physical deduction from the contour plots are given.

2 2. MICROPOLAR CONTINUUM

3 In micropolar theory, the continuum considers the microrotation of the material particles which is added to the Carte-
 4 sian displacements. The material particle of a micropolar continuum experiences displacements and rotations; in a
 5 3D framework there are three displacements and three rotations components, which become two displacements and
 6 one rotation component in the 2D case here considered so that the displacement vector $\{u\}^T = \{u_1 \ u_2 \ \omega\}$ applies.
 7 In the present 2D linearized framework the strain displacement vector is represented as: as $\{\varepsilon\}^T = \{\{\varepsilon_1\} \ \{\varepsilon_2\}\}$,
 8 where $\{\varepsilon_1\} = \{\varepsilon_{11} \ \varepsilon_{22} \ k_1 \ k_2\}$ and $\{\varepsilon_2\} = \{\varepsilon_{12} \ \varepsilon_{21}\}$ and where ε_{11} , ε_{22} , ε_{12} , ε_{21} are the in-plane normal and shear
 9 strains and k_1 , k_2 are the micropolar curvatures. Note that the strain components are not reciprocal, $\varepsilon_{12} \neq \varepsilon_{21}$, The
 10 kinematic compatibility relations can be written as: $\{\varepsilon_1\} = [D_1]\{u\}$ and $\{\varepsilon_2\} = [D_2]\{u\}$

$$[D_1] = \begin{bmatrix} \frac{\partial}{\partial x_1} & 0 & 0 \\ 0 & \frac{\partial}{\partial x_2} & 0 \\ 0 & 0 & \frac{\partial}{\partial x_1} \\ 0 & 0 & \frac{\partial}{\partial x_2} \end{bmatrix}, \quad [D_2] = \begin{bmatrix} \frac{\partial}{\partial x_2} & 0 & 1 \\ 0 & \frac{\partial}{\partial x_1} & -1 \end{bmatrix} \quad (1)$$

11 Note that in general the microrotation, ω , in the micropolar model is different from the local rigid rotation (macro-
 12 rotation), θ , defined as the skew-symmetric part of the gradient of displacement $\theta = \frac{1}{2} \left(\frac{\partial u_2}{\partial x_1} - \frac{\partial u_1}{\partial x_2} \right)$ and the dif-
 13 ference between the two rotations, $\theta - \omega$, defines the strain measure of the relative rotation that corresponds to
 14 the skew-symmetric part of the strain: $\theta = \frac{1}{2} \left(\frac{\partial u_2}{\partial x_1} - \frac{\partial u_1}{\partial x_2} \right)$. When the relative rotation equals zero, $\theta = \omega$ and
 15 $\varepsilon_{12} = \varepsilon_{21} = \frac{1}{2} \left(\frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \right)$, as in the classical continuum, the micropolar continuum becomes a continuum with
 16 constrained rotations (Masiani and Trovalusci (1996); Sokolowski (1972)). In the following, we focus on $\theta - \omega$ as
 17 peculiar strain measure of the micropolarity of the model under study.

18 Analogously, the work-conjugated stress measures of the micropolar model are represented in the vector: $\{\sigma\}^T =$
 19 $\{\{\sigma_1\} \ \{\sigma_2\}\}$ with $\{\sigma_1\}^T = \{\sigma_{11} \ \sigma_{22} \ \mu_1 \ \mu_2\}$ and $\{\sigma_2\}^T = \{\sigma_{12} \ \sigma_{21}\}$, where σ_{ij} for $i, j = 1, 2$ represent the classical
 20 normal and shear stress components, and μ_1 , μ_2 are the microcouples. The stress components are not reciprocal,
 21 $\sigma_{12} \neq \sigma_{21}$, and the couple stress components, μ_1 , μ_2 , have to be introduced in order to satisfy the moment equilibrium
 22 of the micropolar body. From the virtual work principle without introducing the constitutive equations, described

1 below, equilibrium equations in terms of stresses and microcouples can be carried out as $\delta U + \delta V = 0$. So that:

$$\begin{aligned} \delta U &= \int_{\mathcal{V}} \{\delta \varepsilon\}^T \{\sigma\} d\mathcal{V} = \int_{\mathcal{V}} \left(\{\delta \varepsilon_1\}^T \{\sigma_1\} + \{\delta \varepsilon_2\}^T \{\sigma_2\} \right) d\mathcal{V} \\ &= \{\delta u\}^T h \int_{\mathcal{A}} \left([D_1]^T \{\sigma_1\} + [D_2]^T \{\sigma_2\} \right) d\mathcal{A} \end{aligned} \quad (2)$$

2 h being the thickness of the 2D domain which can be considered as unitary for plane strain case, and where the
3 variation of potential of the external loads reads:

$$\delta V = - \int_{\mathcal{V}} \{\delta u\}^T \{f\} d\mathcal{V} - \int_{\mathcal{S}} \{\delta u\}^T \{p\} d\mathcal{S} \quad (3)$$

4 where $\{f\}$ and $\{p\}$ are the vectors of body forces and boundary tractions, respectively. Finally, balance domain
5 equations are given by

$$[D_1]^T \{\sigma_1\} + [D_2]^T \{\sigma_2\} = \{f\} \quad (4)$$

6 and boundary tractions as:

$$\{p\} = \{\hat{p}\} \quad (5)$$

7 where $\{\hat{p}\}$ are stresses and microcouples applied at the boundary.

8 The micropolar constitutive equations take the form:

$$\begin{Bmatrix} \{\sigma_1\} \\ \{\sigma_2\} \end{Bmatrix} = \begin{bmatrix} \mathbb{C}_{11} & \mathbb{C}_{12} \\ \mathbb{C}_{21} & \mathbb{C}_{22} \end{bmatrix} \begin{Bmatrix} \{\varepsilon_1\} \\ \{\varepsilon_2\} \end{Bmatrix} \quad (6)$$

9 and

$$\begin{aligned} \mathbb{C}_{11} &= \begin{bmatrix} A_{1111} & A_{1122} & B_{111} & B_{112} \\ A_{1122} & A_{2222} & B_{221} & B_{222} \\ B_{111} & B_{221} & D_{11} & D_{12} \\ B_{112} & B_{222} & D_{12} & D_{22} \end{bmatrix}, & \mathbb{C}_{12} &= \begin{bmatrix} A_{1112} & A_{1121} \\ A_{2212} & A_{2221} \\ B_{121} & B_{211} \\ B_{122} & B_{212} \end{bmatrix} \\ \mathbb{C}_{21} &= \begin{bmatrix} A_{1112} & A_{2212} & B_{121} & B_{122} \\ A_{1121} & A_{2221} & B_{211} & B_{212} \end{bmatrix}, & \mathbb{C}_{22} &= \begin{bmatrix} A_{1212} & A_{1221} \\ A_{1221} & A_{2121} \end{bmatrix} \end{aligned} \quad (7)$$

FIG. 1: Irregular hexagonal pattern and single tile with geometric parameters $\alpha_1 = 25^\circ$, $l_r = 40$, $\alpha_2 = 45^\circ$, $\alpha_3 = 30^\circ$.

1 Due to hyperelasticity the constitutive matrix is symmetrical: $\mathbb{C}_{12} = \mathbb{C}_{21}^T$. Note that the symbols A_{ijkl} , B_{ijk} and
 2 D_{ij} (for $i, j, k, l = 1, 2$) in definitions (7) can be collected in the matrices \mathbb{A} , \mathbb{B} and \mathbb{D} as in Fantuzzi et al. (2019);
 3 Leonetti et al. (2019). The homogenization technique presented by Trovalusci and Masiani (1999) is used to identify
 4 the constitutive constants of Equation 6 from a given representative volume element (RVE) of elastically interacting
 5 rigid block assemblies.

6 The virtual work principle allows to write the variational statement of the equilibrium for the present micropolar
 7 body $\delta U + \delta V = 0$, where the variation of the strain energy in terms of kinematic parameters reads:

$$\begin{aligned} \delta U &= \int_{\mathcal{V}} \{\delta \varepsilon\}^T \{\sigma\} d\mathcal{V} \\ &= \int_{\mathcal{V}} \left(\{\delta \varepsilon_1\}^T \mathbb{C}_{11} \{\varepsilon_1\} + \{\delta \varepsilon_1\}^T \mathbb{C}_{12} \{\varepsilon_2\} + \{\delta \varepsilon_2\}^T \mathbb{C}_{21} \{\varepsilon_1\} + \{\delta \varepsilon_2\}^T \mathbb{C}_{22} \{\varepsilon_2\} \right) d\mathcal{V} \\ &= \{\delta u\}^T \left[h \int_{\mathcal{A}} \left(\{D_1\}^T \mathbb{C}_{11} \{D_1\} + \{D_1\}^T \mathbb{C}_{12} \{D_2\} + \{D_2\}^T \mathbb{C}_{21} \{D_1\} + \{D_2\}^T \mathbb{C}_{22} \{D_2\} \right) d\mathcal{A} \right] \{u\} \end{aligned} \quad (8)$$

8 In the following, the finite element approximation will be applied directly to the variational principle so that to
 9 equations (3) and (8).

10 3. HEXAGONAL PATTERN

11 The irregular pattern of Figure 1 has been taken from the studies by Scherphuis (2019) which is given by convex
 12 hexagon tilings of type 1 (P6). The correspondent Representative Volume Element (RVE) is given in the same Figure
 13 1 which shows the centroids of the tiles and outward unit normal vectors at the block interfaces used for computing
 14 the constitutive matrix according to the procedure presented by Trovalusci and Masiani (1999).

15 Given a list of input parameters: three angles α_1 , α_2 , α_3 , relative length, l_r , and a tile scale, s , the single hexagon,
 16 tile, can be defined. Using simple geometric formulae the RVE can be defined by translations and mirroring of tiles.

17 The single tile is defined by a parallelogram (skew rectangle) and 2 isosceles triangles with the base attached to
 18 the 2 shortest opposite sides of the parallelogram.

19 With reference to Figure 1, the nodal coordinates of the single tile are given by: $A \equiv (0, 0)$, $B \equiv (l_1, t)$,
 20 $C \equiv (l_1 + l_5, t + l_2/2)$, $D \equiv (l_1, t + l_2)$, $E \equiv (0, l_2)$, $F \equiv (-l_4, l_2/2)$ where $l_2 = s \frac{l_r}{100}$, $l_1 = s - l_2$, $t = l_1 \tan \alpha_1$,
 21 $l_4 = \frac{l_2}{2} \tan \alpha_2$, $l_5 = \frac{l_2}{2} \tan \alpha_3$ and s is a scale parameter. relations among tile side lengths a to f can be easily carried
 22 out using classical geometric relationships.

FIG. 2: Hexagonal patterns given by geometric parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$: a) regular $\alpha_2 = \alpha_3 = 30^\circ$, auxetic $\alpha_2 = \alpha_3 = -20^\circ$ and asymmetric $-\alpha_2 = \alpha_3 = 30^\circ$.

1 By changing the geometric parameters listed above several patterns can be obtained as the ones depicted in Figure
 2 2 which are termed for future reference: regular, hourglass and asymmetric, respectively. They have been obtained by
 3 setting a equilateral geometry (all the sides – a to f – have the same length) with $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1} \cong 63.3975$
 4 and by varying α_2 and α_3 angles as:

5 1. Regular: $\alpha_2 = \alpha_3 = 30^\circ$

6 2. Hourglass: $\alpha_2 = \alpha_3 = -20^\circ$

7 3. Asymmetric: $-\alpha_2 = \alpha_3 = 30^\circ$

8 For the three scales considered the tiles have the following side lengths:

9 • $s = 1 \rightarrow l_1 = 0.3660$

10 • $s = 0.5 \rightarrow l_1 = 0.1830$

11 • $s = 0.25 \rightarrow l_1 = 0.0915$

12 so smaller scales lead to smaller tile sizes and vice versa.

13 4. FINITE ELEMENT IMPLEMENTATION

14 The present finite element framework is based on the previous studies by the authors Fantuzzi et al. (2019); Leonetti
 15 et al. (2019). However, the present implementation has been performed with an in-house MATLAB code as an exten-
 16 sion of the classical 2D plane strain Cauchy continuum, as presented by Ferreira (2008).

17 As it is well-known the classical finite element method enforces an approximation through nodal kinematic
 18 parameters as $\{u\} = \mathbb{N}\{d^e\}$. In this work finite elements with four nodes (Q4) are considered, so that the vector of
 19 nodal parameters reads $\{d^e\}^T = \{u_1^1 \dots u_1^4 \ u_2^1 \dots u_2^4 \ \omega^1 \dots \omega^4\}$, thus each finite element has 12 degrees of freedom
 20 (DOF). Linear shape functions are considered for the present implementation with reduced integration for the micro-
 21 rotation ω due to the fact that un-symmetric strains ε_{12} and ε_{21} will not be given by the same order of quantities. The

1 matrix of the shape functions (of size 3×12) takes the form

$$\mathbb{N} = \begin{bmatrix} \{N\} & \{0\} & \{0\} \\ \{0\} & \{N\} & \{0\} \\ \{0\} & \{0\} & \{N\} \end{bmatrix} \quad (9)$$

2 where $\{N\}$ is the vector of the Lagrangian linear shape functions. Some implementations consider different shape
 3 functions among displacements and micro-rotations (Fantuzzi et al. (2018); Leonetti et al. (2019)) in order to avoid el-
 4 ement locking due to shear strains that have derivative of the planar strains summed with micro-rotation. By inserting
 5 the finite element approximation $\{u\} = \mathbb{N}\{d^e\}$ in the strain energy definition (8) as:

$$\delta U = \{\delta d^e\}^T h \int_{\mathcal{A}} \left(\{B_1\}^T \mathbb{C}_{11} \{B_1\} + \{B_1\}^T \mathbb{C}_{12} \{B_2\} + \{B_2\}^T \mathbb{C}_{21} \{B_1\} + \{B_2\}^T \mathbb{C}_{22} \{B_2\} \right) d\mathcal{A} \{d^e\} \quad (10)$$

6 where $\{B_1\} = \{D_2\}\mathbb{N}$ and $\{B_2\} = \{D_2\}\mathbb{N}$. Thus the element stiffness matrix is:

$$[K^e] = [K_1^e] + [K_2^e] \quad (11)$$

7 where:

$$[K_1^e] = h \int_{\mathcal{A}} \left(\{B_1\}^T \mathbb{C}_{11} \{B_1\} + \{B_1\}^T \mathbb{C}_{12} \{B_2\} + \{B_2\}^T \mathbb{C}_{21} \{B_1\} \right) d\mathcal{A} \quad (12)$$

8

$$[K_2^e] = h \int_{\mathcal{A}} \left(\{B_2\}^T \mathbb{C}_{22} \{B_2\} \right) d\mathcal{A} \quad (13)$$

9 Full integration (2×2 Gauss integration) is performed on $[K_1^e]$, whereas shear strain terms $[K_2^e]$ are integrated using
 10 single point reduced integration.

11 Finally, potential energy (3) becomes:

$$\delta V = -\delta\{d^e\}^T h \int_{\mathcal{A}} \mathbb{N}^T \{f\} d\mathcal{V} - \delta\{d^e\}^T \int_{\mathcal{S}} \mathbb{N}^T \{p\} d\mathcal{S} = -\delta\{d^e\}^T (\{F^e\} + \{P^e\}) \quad (14)$$

12 where $\{F^e\}$ and $\{P^e\}$ are volume and surface force vectors, respectively. In the present work, only surface tractions
 13 are applied so that $\{F^e\} = \{0\}$.

FIG. 3: Current mesh used 16×32 Q4 elements.**1 4.1 Post-computation**

2 Since the present formulation is based on displacements, in order to carry out strain and stress values at discretization
 3 nodes post-computation must be performed. It is well-established that derived quantities should be post-computed at
 4 integration points (Reddy (2004, 2017)). For Q4 elements the integration points used are a 2×2 Gauss-Legendre
 5 grid.

6 Strain and stresses are recovered from the kinematic compatibility and constitutive equations, respectively. The
 7 relative rotation, $\theta - \omega$, is used, as in previous papers (Fantuzzi et al. (2018, 2019); Leonetti et al. (2019); Trovalusci
 8 and Masiani (1999)), for underlining the micropolar effects with respect to the classical continuum. Relative relation

$$\theta - \omega = \left\{ -\frac{1}{2} \frac{\partial}{\partial x_2} \quad \frac{1}{2} \frac{\partial}{\partial x_1} \quad -1 \right\}^{\mathbb{N}} \{d^e\} \quad (15)$$

9 However, field variables are more commonly defined at grid nodes. Thus, extrapolation of each field value is carried
 10 out for Q4 elements using the following formula:

$$\begin{bmatrix} w_1 \\ w_2 \\ w_3 \\ w_4 \end{bmatrix} = \begin{bmatrix} 1 + 0.5\sqrt{3} & -0.5 & 1 - 0.5\sqrt{3} & -0.5 \\ -0.5 & 1 + 0.5\sqrt{3} & -0.5 & 1 - 0.5\sqrt{3} \\ 1 - 0.5\sqrt{3} & -0.5 & 1 + 0.5\sqrt{3} & -0.5 \\ -0.5 & 1 - 0.5\sqrt{3} & -0.5 & 1 + 0.5\sqrt{3} \end{bmatrix} \begin{bmatrix} \hat{w}_1 \\ \hat{w}_2 \\ \hat{w}_3 \\ \hat{w}_4 \end{bmatrix} \quad (16)$$

11 where \hat{w}_i for $i = 1, \dots, 4$ are the extrapolated nodal values whereas w_i for $i = 1, \dots, 4$ are the correspondent internal
 12 field values.

13 Finally, inter-element averaging is applied with equal weights in order to have single values at each nodal point.
 14 This final procedure allows us to have an array of values one for each mesh grid point without intersections.

15 5. NUMERICAL RESULTS

16 A square panel of side $L = 4$ is considered subjected to a top pressure on a limited area $a = L/4$ with a resulting
 17 equivalent concentrated force of $P = 10^3$ pointing downward. The panel is clamped at the bottom. A regular squared
 18 FE mesh of 16×32 elements (by taking advantage of problem symmetry) is considered and depicted in Figure 3.

19 In this section the numerical results for the three cited geometries are presented. In particular, the contour plots of

FIG. 4: Regular hexagonal pattern RVE with geometric parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = 30^\circ$ at different scale $s = 1, 0.5, 0.25$ from left to right respectively.

1 displacements, u_1 (horizontal) and u_2 (vertical), stresses, σ_{11} (horizontal), σ_{22} (vertical), and relative rotation $\theta - \omega$
2 are shown for each geometry.

3 The homogenization procedure follows the approach described by Trovalusci and Masiani (1999), where the
4 adopted spring stiffness at the elastic joint interfaces are:

$$K_T = \begin{bmatrix} 785 & 0 \\ 0 & 780 \end{bmatrix}, \quad K_R = 0 \quad (17)$$

5 where K_T and K_R are the stiffnesses of translational and rotational springs. Energetic equivalence is used to
6 carry out rotational stiffnesses as:

$$k_n = K_{T(1,1)} \frac{d}{2}, \quad k_r = k_n \frac{d^2}{2} = K_{T(1,1)} \frac{d^3}{4} \quad (18)$$

7 where d is the current interface length between two rigid particles for which the interactive stiffness is computed.

8 For all three different scales, stiffness data (K_T and K_R , and consequently k_n and k_r) remain unchanged. The
9 ratios between tile length and panel width l_1/L are: 10.93, 21.86, 43.72 by decreasing scale, respectively. The consti-
10 tutive matrices are presented below in compact and extended matrix form:

$$\mathbf{C} = \begin{bmatrix} \mathbf{A} & \mathbf{B} \\ \mathbf{B}^T & \mathbf{D} \end{bmatrix} = \begin{bmatrix} A_{1111} & A_{1122} & A_{1112} & A_{1121} & B_{111} & B_{112} \\ & A_{2222} & A_{2212} & A_{2221} & B_{221} & B_{222} \\ & & A_{1212} & A_{1221} & B_{121} & B_{122} \\ & & & A_{2121} & B_{211} & B_{212} \\ & & & & D_{11} & D_{12} \\ \text{sym} & & & & & D_{22} \end{bmatrix} \quad (19)$$

11 5.1 Orthotetragonal hexagons

12 The aforementioned regular hexagons can be carried out from P6 geometry Scherphuis (2019) by setting: $l_r =$
13 63.3975, $\alpha_1 = 0^\circ$, $\alpha_2 = \alpha_3 = 30^\circ$. The results for the three scales $s = 1, 0.5, 0.25$ are depicted in Figure 4, where
14 tile centers and link orientations are shown for calculation purposes (Trovalusci and Masiani (1999)).

1 The correspondent constitutive matrices appear to be orthotetragonal (symmetric with respect to a 90° rotation)
 2 as:

$$\mathbb{C}_{\text{ortho}}^{(s=1)} = \begin{bmatrix} 496.8776 & 0.7925 & 0 & 0 & 0 & 0 \\ 0.7925 & 496.8776 & 0 & 0 & 0 & 0 \\ 0 & 0 & 495.2927 & 0.7925 & 0 & 0 \\ 0 & 0 & 0.7925 & 495.2927 & 0 & 0 \\ 0 & 0 & 0 & 0 & 33.3376 & 0 \\ 0 & 0 & 0 & 0 & 0 & 33.3376 \end{bmatrix} \quad (20)$$

$$\mathbb{C}_{\text{ortho}}^{(s=0.5)} = \begin{bmatrix} 248.4388 & 0.3962 & 0 & 0 & 0 & 0 \\ 0.3962 & 248.4388 & 0 & 0 & 0 & 0 \\ 0 & 0 & 247.6463 & 0.3962 & 0 & 0 \\ 0 & 0 & 0.3962 & 247.6463 & 0 & 0 \\ 0 & 0 & 0 & 0 & 4.1672 & 0 \\ 0 & 0 & 0 & 0 & 0 & 4.1672 \end{bmatrix} \quad (21)$$

$$\mathbb{C}_{\text{ortho}}^{(s=0.25)} = \begin{bmatrix} 124.2194 & 0.1981 & 0 & 0 & 0 & 0 \\ 0.1981 & 124.2194 & 0 & 0 & 0 & 0 \\ 0 & 0 & 123.8232 & 0.1981 & 0 & 0 \\ 0 & 0 & 0.1981 & 123.8232 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0.5209 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0.5209 \end{bmatrix} \quad (22)$$

3 It can be noted that in such orthotetragonal assemblies $A_{1111} = A_{2222}$ and $A_{1212} = A_{2121}$ as well as $D_{11} = D_{22}$,
 4 moreover there is no coupling between normal and shear stresses and at the same time between microcouples ($D_{12} =$
 5 0) and $\mathbb{B} = 0$ thus no elastic coefficients relating stresses and strains and coefficients relating microcouples and
 6 curvatures is expected. A very small Poisson effect is also shown. The ratios between elastic coefficients relating
 7 stresses and strains between different scales is 2, whereas between coefficients relating microcouples and curvatures
 8 is 8 ($\mathbb{A}^{(s=1)} = 2\mathbb{A}^{(s=0.5)} = 4\mathbb{A}^{(s=0.25)}$ and $\mathbb{D}^{(s=1)} = 8\mathbb{D}^{(s=0.5)} = 64\mathbb{D}^{(s=0.25)}$).

9 The numerical results obtained are shown in Figures 5-9 for the three mentioned scales. We can observe that
 10 in such a case of localized load the micropolar model is able to better distribute the load depending on the material

FIG. 5: Q4 formulation. Horizontal displacement u_1 for regular hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 6: Q4 formulation. Vertical displacement u_2 for regular hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

1 internal length related to the tile size. In particular, the vertical displacement, u_2 , tends to concentrate on the central
 2 part of the wall for smaller scales with more remarked values with respect to larger scales (Figure 6). Due to vertical
 3 pressure the body also tends to move horizontally, u_1 . Figure 5 displays such effect which is more evident for smaller
 4 scales.

5 Horizontal pressures, σ_{11} , (Figure 7) are almost negligible for larger scales. For small scales these stresses con-
 6 centrate in the area below the applied load. The same effect is more clear for the vertical stress, σ_{22} , where a distinct
 7 flux of stress is clearly visible for smaller scales (Figure 8).

8 Finally, the relative rotation, $\theta - \omega$, plot (Figure 9) shows a very small micropolar effect as expected since
 9 orthotetragonal materials are known to be very close to Cauchy continua, as already discussed in (Trovalusci and
 10 Masiani, 1999).

11 5.2 Hourglass hexagons

12 By keeping the symmetry of the single tile and equilateral tile segments, the inner angles $\alpha_2 = \alpha_3 = -20^\circ$ are set to
 13 a negative value. Graphical representation of the three cells with different scale is given in Figure 10.

14 This leads to an auxetic behavior of the homogenized material, with negative Poisson coefficient ($A_{1122} < 0$), for
 15 all three cases considered as it is shown by the constitutive matrices given below:

$$\mathbb{C}_{\text{aux}}^{(s=1)} = \begin{bmatrix} 209.2 & -0.5 & 0 & 0 & 0 & 0 \\ -0.5 & 1059.7 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1056.8 & -0.5 & 0 & 0 \\ 0 & 0 & -0.5 & 208.2 & 0 & 0 \\ 0 & 0 & 0 & 0 & 11.9 & 0 \\ 0 & 0 & 0 & 0 & 0 & 67.7 \end{bmatrix} \quad (23)$$

FIG. 7: Q4 formulation. Horizontal stress σ_{11} for regular hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 8: Q4 formulation. Vertical stress σ_{22} for regular hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 9: Q4 formulation. Relative rotation $\theta - \omega$ for regular hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

$$\mathbb{C}_{\text{aux}}^{(s=0.5)} = \begin{bmatrix} 104.6165 & -0.2710 & 0 & 0 & 0 & 0 \\ -0.2710 & 529.8742 & 0 & 0 & 0 & 0 \\ 0 & 0 & 528.3768 & -0.2710 & 0 & 0 \\ 0 & 0 & -0.2710 & 104.1057 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1.4892 & 0 \\ 0 & 0 & 0 & 0 & 0 & 8.4671 \end{bmatrix} \quad (24)$$

$$\mathbb{C}_{\text{aux}}^{(s=0.25)} = \begin{bmatrix} 52.3083 & -0.1355 & 0 & 0 & 0 & 0 \\ -0.1355 & 264.9371 & 0 & 0 & 0 & 0 \\ 0 & 0 & 264.1884 & -0.1355 & 0 & 0 \\ 0 & 0 & -0.1355 & 52.0529 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0.1861 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1.0584 \end{bmatrix} \quad (25)$$

1 The ratio between elastic coefficients relating stresses and strains and coefficients relating microcouples and
 2 curvatures results as in the previous case ($\mathbb{A}^{(s=1)} = 2\mathbb{A}^{(s=0.5)} = 4\mathbb{A}^{(s=0.25)}$; $\mathbb{D}^{(s=1)} = 8\mathbb{D}^{(s=0.5)} = 64\mathbb{D}^{(s=0.25)}$). In
 3 addition, $A_{1111} \cong A_{2121}$, $A_{2222} \cong A_{1212}$ where the former are circa 5 times the latter and $5.69D_{11} \cong D_{22}$.

4 Numerical results are given in Figures 11-15. The auxetic effect can be clearly seen in the horizontal displace-
 5 ments u_1 plot (Figure 11) where the particles move on the right – towards the panel central area – this effect is quite
 6 small for larger particles and increases rapidly for smaller scales.

7 Vertical displacements, u_2 , (Figure 12), stresses, σ_{11} , σ_{22} (Figures 13, 14), have comparable behavior with respect
 8 to the correspondent orthotetragonal cases but they have smaller intensity. It is remarked that the vertical stress is
 9 better diffused within the panel more in the case of larger size, in fact remarkable stress values are shown in Figure
 10 14 which are higher than the ones of the correspondent orthotetragonal case 8.

FIG. 10: Regular hexagonal pattern RVE with geometric parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = -20^\circ$ at different scale $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 11: Q4 formulation. Horizontal displacement u_1 for auxetic hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = -20^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 12: Q4 formulation. Vertical displacement u_2 for auxetic hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = -20^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

1 Relative rotation, $\theta - \omega$ (Figure 15), shows a clear micropolar effect which increases when the scale s is de-
2 creased, this means that such a micropolar effect is stronger for smaller particles.

3 5.3 Chiral hexagons

4 The present geometry is obtained considering equilateral hexagons with inner angles $-\alpha_2 = \alpha_3 = 30^\circ$. Graphical
5 representation of the three cells with different scale is given in Figure 16.

6 The present geometrical selection leads to the following homogenized materials for the three given scales

$$\mathbb{C}_{\text{chi}}^{(s=1)} = \begin{bmatrix} 331.2517 & 0 & 0 & 0 & 0 & 0 \\ 0 & 745.3164 & 0 & 0 & 0 & 45.5400 \\ 0 & 0 & 742.9390 & 0 & 0 & 0 \\ 0 & 0 & 0 & 330.1951 & 0 & 0 \\ 0 & 0 & 0 & 0 & 22.2250 & 0 \\ 0 & 45.5400 & 0 & 0 & 0 & 50.0064 \end{bmatrix} \quad (26)$$

$$\mathbb{C}_{\text{chi}}^{(s=0.5)} = \begin{bmatrix} 165.6259 & 0 & 0 & 0 & 0 & 0 \\ 0 & 372.6582 & 0 & 0 & 0 & 11.3850 \\ 0 & 0 & 371.4695 & 0 & 0 & 0 \\ 0 & 0 & 0 & 165.0976 & 0 & 0 \\ 0 & 0 & 0 & 0 & 2.7781 & 0 \\ 0 & 11.3850 & 0 & 0 & 0 & 6.2508 \end{bmatrix} \quad (27)$$

FIG. 13: Q4 formulation. Vertical stress σ_{11} for auxetic hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = -20^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 14: Q4 formulation. Vertical stress σ_{22} for auxetic hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = -20^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 15: Q4 formulation. Relative rotation $\theta - \omega$ for auxetic hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $\alpha_2 = \alpha_3 = -20^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

$$\mathbb{C}_{\text{chi}}^{(s=0.25)} = \begin{bmatrix} 82.8129 & 0 & 0 & 0 & 0 & 0 \\ 0 & 186.3291 & 0 & 0 & 0 & 2.8462 \\ 0 & 0 & 185.7347 & 0 & 0 & 0 \\ 0 & 0 & 0 & 82.5488 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0.3473 & 0 \\ 0 & 2.8462 & 0 & 0 & 0 & 0.7813 \end{bmatrix} \quad (28)$$

1 Note that in this case a coupling between elastic coefficients relating stresses and strains and coefficients relating
 2 microcouples and curvatures occurs, implying coupling between normal stress σ_{22} and curvature k_2 and microcouple
 3 μ_2 and normal strain ε_2 . No other coupling takes place (also no Poisson effect is observed). Such behavior is due
 4 to the selection of equal and opposite angles $-\alpha_2 = \alpha_3 = 30^\circ$ which break material symmetry and neutralize
 5 lateral contraction/expansion. The ratio between coefficients relating stresses and strains and coefficients relating
 6 microcouples and curvatures is the same as in the previous micro-structures ($\mathbb{A}^{(s=1)} = 2\mathbb{A}^{(s=0.5)} = 4\mathbb{A}^{(s=0.25)}$)
 7 and $\mathbb{D}^{(s=1)} = 8\mathbb{D}^{(s=0.5)} = 64\mathbb{D}^{(s=0.25)}$. In addition, since $\mathbb{B} \neq 0$, $\mathbb{B}^{(s=1)} = 4\mathbb{B}^{(s=0.5)} = 16\mathbb{B}^{(s=0.25)}$. Similarly
 8 to the auxetic configuration, $A_{1111} \cong A_{2121}$, $A_{2222} \cong A_{1212}$, where the former are circa 2.25 times the latter and
 9 $2.27D_{11} \cong D_{22}$.

10 The numerical results are given in Figures 17-21.

11 Since no Poisson effect is given in the constitutive equations negligible horizontal displacement u_1 (Figure 17)
 12 and stress σ_{11} (Figure 19) are observed. Vertical displacement u_2 (Figure 18) and stress σ_{22} (Figure 20) have a
 13 similar trend with respect to the other cases but closer to the auxetic one. The coupling in the present microstructure
 14 with $B_{222} \neq 0$ gives a lower relative rotation effect than the one in the auxetic material but its variation is not uniform
 15 close to the symmetry axis, because the vertical pressure σ_{22} is coupled with the vertical curvature k_2 . This effect can
 16 be clearly observed for any scale s in Figure 21.

FIG. 16: Regular hexagonal pattern RVE with geometric parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $-\alpha_2 = \alpha_3 = 30^\circ$ at different scale
 $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 17: Q4 formulation. Horizontal displacement u_1 for asymmetric hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $-\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 18: Q4 formulation. Vertical displacement u_2 for asymmetric hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $-\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

1 6. CONCLUSIONS

2 The present work investigates the static behavior of materials with three types of hexagonal microstructures described
 3 as equivalent micropolar media. Such micro-structures are typical of polycrystals with thin interfaces such as Alumina
 4 (Al_2O_3), Zirconia (ZrO_2), Zinc Oxide (ZnO) or Tungsten-Carbide (WC) just to cite a few. The three selected patterns
 5 provides orthotetragonal, auxetic and chiral material symmetries and each one showed some peculiarities and some
 6 interesting outcomes related to the micropolar behavior. Assemblies of regular hexagons have an orthotetragonal
 7 behavior and it has been shown that their homogenized behavior is close to the behavior of classical elastic bodies
 8 Masiani and Trovalusci (1996); Trovalusci and Masiani (2005). On the contrary, the other configurations showed
 9 strong nonlocal effects, related to the internal material size, which brought a larger stress diffusion within the body
 10 and reduced displacements. All these aspects could be underlined mostly because a micropolar continuum has extra
 11 rotational degrees of freedom with respect to classical continuum. If the coupling effect among classical and micro-
 12 polar quantities wanted to be observed elastic interfaces among particles have to be as less symmetric as possible.
 13 Other geometries and interfaces configurations will be considered in future works in order to better understand the
 14 prediction of micropolar effects in microstructured materials.

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FIG. 19: Q4 formulation. Vertical stress σ_{11} for asymmetric hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $-\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 20: Q4 formulation. Vertical stress σ_{22} for asymmetric hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $-\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

FIG. 21: Q4 formulation. Relative rotation $\theta - \omega$ for asymmetric hexagonal patterns with parameters $\alpha_1 = 0^\circ$, $l_r = \frac{100}{1/\sqrt{3}+1}$, $-\alpha_2 = \alpha_3 = 30^\circ$ and scales $s = 1, 0.5, 0.25$ from left to right respectively.

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